TECHNICAL MANUAL

Operator's and Crewmember's Checklist

ARMY MODEL RC-12D AIRCRAFT

NSN 1510-01-087-9129 (EIC: SRC)

Pilot's Checklist

This copy is a reprint which includes current pages from Change 4.

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*TM 55-1510-219-CL, dated 31 May 1991 supersedes TM 55-1510-219-CL, dated 25 May 1985, including all changes.

> HEADQUARTERS, DEPARTMENT OF THE ARMY

> > 31 May 1991

CHANGE 5

This change was not available at the time of the CD release Please refer to your paper or microfiche copy as appropriate.

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URGENT

TM 55-1510-219-CL C 4

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TECHNICAL MANUAL Operator's and Crewmember's Checklist ARMY MODEL RC-12D AIRCRAFT

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TECHNICAL MANUAL Operator's and Crewmember's Checklist ARMY MODEL RC-12D AIRCRAFT

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TECHNICAL MANUAL

Operator's and Crewmember's Checklist

ARMY MODEL RC-12D AIRCRAFT

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TM 55-1510-219-CL C 1

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TECHNICAL MANUAL

Operator's and Crewmember's Checklist

ARMY MODEL RC-12D AIRCRAFT NSN 1510-01-087-9129 (EIC: SRC)

Pilot's Checklist

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TM 55-1510-219-CL

GENERAL INFORMATION AND SCOPE

SCOPE. This checklist contains the operator's and crewmembers checks to be accomplished during normal and emergency operations.

GENERAL INFORMATION. The checklist consists of three parts: normal procedures, emergency procedures, and performance data. Normal procedures consist of the procedures required for normal flight and those required for "Before Landing". The normal procedures portion will be subdivided to include the before landing checks of Chapter 8 of the Operator's manual. Emergency procedures are subdivided into 7 classifications as follows: engine, propeller. (PROP.), fire, fuel, electrical (ELECT.), landing and ditching (LDG/DTCH) and flight controls (FLT CONT.). Performance data consists of performance checks.

NOTE

This checklist does not replace the amplified version of the procedures in the operator's manual (TM 55-1510-219-10), but is a condensed version of each procedure.

NORMAL PROCEDURES PAGES. The contents of the normal procedures of this manual are a condensation of the amplified checklist appearing in the normal procedures or crew duties portion of the operator's manual.

EMERGENCY PROCEDURES PAGES. The requirements in this section of the condensed checklist (CL) are identical to those for the normal procedures, except that the information is drawn from the amplified procedures in the emergency procedures portion of the operator's manual. Immediate action items are underlined for your reference and shall be committed to memory. Symbols Preceding Numbered Steps

- Indicates the performance of these steps is mandatory for all "Thru Flights"
- N Indicates performance of step is mandatory for "Night Flights"
 - Indicates a detailed procedure for this step is included in the Performance Checks section, located at the back of the checklist
- Indicates mandatory check for "Instrument Flight"
- 0 Indicates if installed

Immediate action emergency items are underlined

REPORTING ERRORS AND RECOMMENDING IMPROVEMENTS

You can help improve this manual. If you find any mistakes or if you know of a way to improve the procedures, please let us know. Mail your letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms) or DA Form 2028-2, located in the back of the applicable Aircraft Operator's Manual, directly to: Commander, U.S. Army Aviation and Missile Command, ATTN: AMSAM-MMC-LS-LP, Redstone Arsenal, AL 35898-5230. You may also send in your comments electronically to our e-mail address at <ls-lp@redstone.army.mil> or by fax at (205) 842-6546 or DSN 788-6546. Instructions for sending in an electronic 2028 may be found at the back of the appropriate Aircraft Operator's Manual. A reply will be furnished directly to you.

NORMAL PROCEDURES

BEFORE EXTERIOR CHECK

- 1. Publications Check DA Forms 2408-12,-13,-14, -18, DD Form 365-4, locally required forms and publications, and availability of operator's manual (-10) and checklist (-CL).
- *2. Oxygen system Check as required.
- 3. Flight controls Unlock and check.
- 4. Parking brake Set.
- 5. Manual trim Zero.
- 6. Gear- DN.
- 7. Ice vanes IN.
- 8. Overhead panel switches and circuit breakers -Set.
- 9. AC and DC GPU's As required.
- 10. External power advisory lights As required.
- 11. Keylock switch -ON.
- *12. Fuel pumps and crossfeed operation Check.
- 13. DC power Check (24 VDC minimum for battery, 28 maximum for GPU starts).
- 14. Lighting systems Check.
- *15. Anti-ice systems Check.
- *16. Annunciator panels Test as required.
- *17. Fire protection system Check.
- *18. INS alignment As required.
- *19. Electric elevator trim and autopilot/flight director operation Check.

- *20. Avionics Check.
- 21. Flaps Check.
- 22. Battery switch As required.
- 23. Toilet Check.
- 24. Emergency equipment Check.
- 25. Mission equipment and circuit breakers Check and set.
- 26. Parachutes Check (as required).

FUEL SAMPLE

*1. Check collective fuel sample.

LEFT WING, AREA 1

- 1. General condition Check.
- 2. Flaps Check.
- 3. Fuel sump drains (3) Check.
- 4. Controls and trim tab Check.
- 5. Static wicks Check.
- 6. Wing pod, navigation lights and antennas (2) -Check.
- 7. Recognition light Check.
- 8. Outboard antenna set Check.
- *9. Main tank fuel and cap Check.
- 10. Outboard wing fuel vent Check.
- 11. Outboard deice boot Check.
- 12. Stall warning vane Check.
- *13. Tiedown Released.
- 14. Inboard dipole antenna set Check.
- 15. Wing ice light Check.

- 16. AC GPU access door Secure.
- 17. Recessed and heated fuel vents Check.
- 18. Inverter inlet and exhaust louvers Check.

LEFT MAIN LANDING GEAR

- *1. Tires Check.
- 2. Brake assembly Check.
- *3. Shock strut Check.
- 4. Torque knee Check.
- 5. Safety switch Check.
- 6. Fire extinguisher pressure Check.
- 7. Wheel well, doors, and linkage Check.
- 8. Fuel sump drains (forward) Check.

LEFT ENGINE AND PROPELLER

- *1. Engine oil Check.
- 2. Engine compartment left side Check.
- *3. Left cowl locks Locked.
- 4. Left exhaust stack Check.
- *5. Propeller blades and spinner Check.
- "6. Engine air Inlets and ice vane Check.
- 7. Bypass door Check.
- *8. Right cowl locks Locked.
- 9. Right exhaust stack Check.
- 10. Engine compartment, right side Check.

CENTER SECTION, LEFT SIDE

- 1. Heat exchanger inlet and outlet Check.
- 2. Auxiliary tank fuel sump drain Check.

- 3. Deice boot Check.
- *4. Auxiliary tank fuel gage and cap Check.
- 5. Monopole antenna Check.

FUSELAGE UNDERSIDE

- *1. General condition Check.
- 2. Antennas Check.

NOSE SECTION, AREA 2

- 1. Free air temperature probe Check.
- 2. Avionics door, left side Check.
- 3. Air conditioner exhaust Check.
- 4. Wide band data link antenna pod Check.
- 5. Wheel well Check.
- 6. Doors and linkage Check.
- 7. Nose gear turning stop Check.
- *8. Tire Check.
- *9. Shock strut Check.
- 10. Torque knee Check.
- 11. Shimmy damper and linkage Check.
- 12. Landing and taxi lights Check.
- 13. Pitot tubes Check.
- 14. Radome Check.
- 15. Windshields and wipers Check.
- 16. Air conditioner inlet Check.
- 17. Avionics door, right side Check.

RIGHT WING CENTER SECTION

1. Deice boot - Check.

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- 2. Battery access panel Secure.
- 3. Battery vents Check.
- *4. Auxiliary tank fuel and cap Check.
- 5. Battery compartment drain Check.
- 6. Battery ram air intake Check.
- 7. INS temperature probe Check.
- 8. Auxiliary tank fuel sump drain Check.
- 9. Heat exchanger outlet and inlet Check.
- 10. Monopole antenna Check.

RIGHT ENGINE AND PROPELLER

- *1. Engine oil Check.
- 2. Engine compartment, left side Check.
- *3. Left cowl locks Locked.
- 4. Left exhaust stack Check.
- *5. Propeller blades and spinner Check.
- *6. Engine air inlets and ice vane Check.
- 7. Bypass door Check.
- *8. Right cowl locks Locked.
- 9. Right exhaust stack Check.
- 10. Engine compartment, right side Check.

RIGHT MAIN LANDING GEAR

- 1. Fuel sump drains (forward) Check.
- *2. Tires Check.
- 3. Brake assembly Check.
- *4. Shock strut Check.
- 5. Torque knee Check.
- 6. Safety switch Check.

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- 7. Fire extinguisher pressure Check.
- 8. Wheel well, doors, and linkage Check.

RIGHT WING, AREA 3

- 1. Recessed and heated fuel vents Check.
- 2. Inverter inlet and exhaust louvers Check.
- 3. GPU access door Secured.
- 4. Inboard dipole antenna set Check.
- 5. Wing ice light Check.
- 6. Outboard deice boot Check.
- *7. Tiedown Released.
- *8. Main tank fuel and cap Check.
- 9. Outboard wing fuel vent Check.
- 10. Outboard antenna set Check.
- 11. Recognition light Check.
- 12. Wing pod, navigation lights and antennas (2) -Check.
- 13. Static wicks Check.
- 14. Controls Check.
- 15. Fuel sump drains (3) Check.
- 16. Flaps Check.
- 17. Chaff dispenser Check.
- *18. General condition Check.

FUSELAGE, RIGHT SIDE, AREA 4

- *1. General condition Check.
- 2. Flare/Chaff dispenser Check.
- 3. Emergency light Check.
- 4. Beacon Check.

- 5. Aft access door Check.
- 6. Oxygen filler door Check.
- 7. Static ports Check.
- 8. ASE antennas (2) Check.
- 9. Emergency locator transmitter ARMED.
- 10. Emergency locator transmitter antenna Check.

EMPENNAGE, AREA 5

- 1. Vertical stabilizer, rudder, and trim tab Check.
- 2. Antennas Check.
- 3. Deice boots Check.
- 4. Horizontal stabilizer, and elevator Check.
- 5. Elevator trim tab Verify "0' (neutral) position.
- 6. Static wicks (16) Check.
- 7. Position and beacon lights Check.
- 8. Rotating boom dipole antenna Check.
- 9. Wide band data link antenna pod Check.

FUSELAGE, LEFT SIDE, AREA 6

- *1. General condition Check.
- 2. Static ports Check.
- 3. ASE antennas (2) Check.
- 4. Emergency light Check.
- 5. Cabin door Check.
- 6. Fuselage top side Check.

***INTERIOR CHECK**

- 1. Cargo/loose equipment Check.
- 2. Cabin door Locked and checked.

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- 3. Cargo door Locked and checked.
- 4. Emergency exit Check.
- 5. Mission cooling ducts Check.
- 6. Flare/Chaff dispenser preflight test Completed.
- 7. Crew briefing As required.

BEFORE STARTING ENGINES

- *1. Parking brake Set.
- 2. Magnetic compass Check.
- *3. Pedestal controls Set.
- 4. Lower console switches Set.
- 5. Gear alternate engage and ratchet handles -Stowed.
- 6. Free air temperature gage Check.
- 7. Instrument panel Check.
- 8. Prop sync switch OFF.
- 9. Mission panel switches and circuit breakers -Set.
- 10. Subpanels Check.
- 11. AC and DC GPU's As required.
- 12. External power advisory lights As required.
- *13. Battery ON.
- 14. DC power Check.

*FIRST ENGINE START (BATTERY START)

- 1. Avionics master switch OFF.
- 2. Exterior light switches As required.

- 3. Ignition and engine start switch ON.
- 4. Condition lever (after N1 RPM stabilizes, 12% minimum) LOW IDLE.
- 5. TGT and N1 Monitor (TGT 1000°C maximum, N₁ 52% minimum).
- 6. 0il pressure Check (60 PSI minimum).
- 7. Ignition and engine start switch OFF after TGT stabilized.
- 8. Condition lever- HIGH IDLE.
- 9. Generator switch RESET, then ON.

SECOND ENGINE START (BATTERY START)

- 1. First engine generator load 50% or less.
- 2. Ignition and engine start switch ON.
- 3. Condition lever (after N1 RPM passes 12% minimum) LOW IDLE.
- 4. TGT and N1 Monitor (TGT 1000°C maximum, N₁ 52% minimum).
- 5. Oil pressure Check (60 PSI minimum).
- 6. Ignition and engine start switch OFF.
- 7. Battery charge light ON.
- 8. Second engine generator RESET, then ON.
- 9. Inverter switches ON.
- 10. Condition levers As required.

ABORT START

- 1. Condition lever FUEL CUTOFF.
- 2. Ignition and engine start switch STARTER ONLY.

- 3. TGT- Monitor.
- 4. Ignition and engine start switch OFF after TGT stabilized.

ENGINE CLEARING

- 1. Condition lever FUEL CUTOFF.
- 2. Ignition and engine start switch OFF.
- 3. Ignition and engine start switch STARTER ONLY.
- 4. Ignition and engine start switch OFF.

FIRST ENGINE START (GPU START)

- 1. INS As required.
- 2. Avionics master switch As required.
- 3. Exterior light switches As required.
- 4. Ignition and engine start switch ON.
- 5. Condition lever (after N1, RPM stabilizes, 12% minimum) LOW IDLE.
- 6. TGT and N1 Monitor (TGT 1000°C maximum, N1 52% minimum).
- 7. Oil pressure Check (60 PSI minimum).
- 8. Ignition and engine start switch OFF after TGT stabilized.
- 9. Condition lever As required.
- 10. DC GPU disconnect As required.
- 11. Generator switch (GPU disconnected) RESET, then ON.
- 12. Condition lever HIGH IDLE.

SECOND ENGINE START (GPU START)

- 1. Ignition and engine start switch ON.
- 2. Condition lever (after N₁ RPM passes, 12% minimum) LOW IDLE.
- 3. TGT and N_1 Monitor (TGT 1000°C maximum, N_1 52% minimum).
- 4. Oil pressure Check (60 PSI minimum).
- 5. Ignition and engine start switch OFF after TGT stabilized.
- 6. Propeller levers FEATHER.
- 7. GPU Disconnect.
- 8. Propellers levers HIGH RPM.
- 9. Generator switches RESET, then ON.
- 10. Aircraft inverter switches ON, check INVERTER lights off.
- 11. Condition levers As required.

BEFORE TAXIING

- 1. Brake deice As required.
- *2. Cabin temperature and mode Set.
- 3. AC/DC power Check.
- 4. Avionics master power switch ON as required.
- 5. Mission panel Set.
- 6. Electric elevator trim and autopilot/flight director operation Check.
- 7. Avionics Check.
- 8. Flaps Check.
- 9. Altimeters Check and set.

*TAXIING

- 1. Brakes Check.
- 2. Flight instruments Check.
- 3. Mission control panel Set as required.

ENGINE RUNUP

- 1. Propeller manual feathering Check.
- ★2. Autofeather Check.
- ★3. Overspeed Governors Check.
- ★4. Primary governors Check.
- ★5. Ice vanes Check.
- \star 6. Anti-ice and deice systems Check.
 - 7. Beacon As required.
- ★8. Pneumatic pressure Check.
- ★9. Pressurization system Check.
- 10. Windshield anti-ice As required.

*BEFORE TAKEOFF

- 1. Autofeather switch ARM.
- 2. Bleed air valves As required.
- 3. Ice and rain switches As required.
- 4. Fuel panel Check.
- 5. Flight and engine instruments Check.
- 6. Cabin controller Set
- 7. Annunciator panels Check.
- 8. Propeller levers HIGH RPM.
- 9. Friction locks Set
- 10. Flaps As required.
- 11. Trim-Set.

- 12. Avionics Set
- 13. Flight controls Check.
- 14. Departure briefing Complete.

*LINE UP

- 1. Transponder- As required.
- 2. Engine auto ignition switch ARM.
- 3. Power stabilized Check.
- 4. Condition levers LOW IDLE.
- 5. Lights As required.
- 6. Mission control panel Set.

AFTER TAKEOFF

- 1. Gear- UP.
- 2. Flaps UP.
- 3. Landing lights OFF.
- ④ Windshield anti-ice- As required.

CLIMB - MAXIMUM RATE

- 1. Climb power Set.
- 2. Propeller sync As required.
- 3. Autofeather- As required.
- 4. Yaw damp As required.
- 5. Cabin pressurization Check.
- 6. Wings and nacelles Check.
- 7. ASE-As required.

CRUISE

1. Power- Set.

- 2. Wings and nacelles Check.
- 3. Ice and rain switches As required.
- 4. Auxiliary fuel gages Monitor.
- 5. Altimeters Check.
- 6. Engine instrument indications Noted.
- 7. Recognition lights As required.

DESCENT - MAX RATE (CLEAN)

- 1. Power levers IDLE.
- 2. Propeller levers HIGH RPM.
- 3. Flaps UP.
- 4. Gear UP.
- 5. Airspeed V_{mo}.
- 6. Cabin pressurization Set.
- 7. Ice and rain switches As required.
- 8. Recognition lights As required.

DESCENT - MAX RATE (LANDING CONFIGURATION)

- 1. Power levers IDLE.
- 2. Propeller levers HIGH RPM.
- 3. Flap switch APPROACH.
- 4. Gear switch DN.
- 5. Airspeed 184 KIAS.
- 6. Cabin pressurization Set.
- 7. Ice and rain switches As required.
- 8. Recognition lights As required.

DESCENT-ARRIVAL

- 1. Cabin pressurization Set.
- 2. Ice and rain switches As required.
- ③. Windshield anti-ice- As required.
 - 4. Lights- ON.
 - 5. Altimeters Set.
 - 6. ASE As required.

BEFORE LANDING

- 1. Prop sync switch- OFF.
- 2. Autofeather switch ARM.
- 3. Propeller levers As required.
- 4. Flap switch (below 202 KIAS) APPROACH.
- 5. Gear switch (below 184 KIAS) DN.
- 6. Rotating boom dipole antenna Check stowed.
- 7. Landing lights As required.
- 8. Brake deice As required.

LANDING

- 1. Autopilot and yaw damp Disengaged.
- 2. Gear down lights Check
- 3. Propeller levers HIGH RPM.

GO-AROUND

- 1. Power- Maximum allowable.
- 2. Gear UP.
- 3. Flaps UP.
- 4. Landing lights OFF.
- 5. Climb power Set.

C1 N-15

6. Yaw damp - As required.

AFTER LANDING

- 1. Condition levers As required.
- 2. Engine auto ignition switch OFF.
- 3. Ice and rain switches OFF.
- 4. Flaps UP.
- 5. Avionics As required.
- 6. Lights As required.
- 7. Mission control panel Set.

ENGINE SHUTDOWN

- 1. Brake deice OFF.
- 2. Parking brake Set.
- 3. Landing/taxi lights OFF.
- 4. Overhead floodlight As required.
- 5. Cabin temperature mode switch OFF.
- 6. Autofeather switch OFF.
- 7. Vent and aft vent blower switches AUTO.
- 8. INS OFF.
- 9. Mission equipment OFF, as required.
- 10. Inverter switches OFF.
- 11. Battery condition Check as required.
- 12. Avionics master switch OFF.
- 13. TGT Check.
- 14. Propeller levers FEATHER.
- 15. Condition levers FUEL CUTOFF.
- 16. Exterior lights OFF.

- 17. Master panel lights OFF.
- 18. Master switch OFF.
- 19. Keylock switch OFF.
- 20. Oxygen system As required.

SEFORE LEAVING AIRCRAFT

- 1. Wheels Chocked
- 2. Parking brake As required
- 3. Flight controls Locked.
- 4. Fuel pumps Set.
- 5. Emergency exit lock As required.
- 6. Mode 4 As required.
- 7. Aft cabin light OFF.
- 8. Door light OFF.
- 9. Walk-around Inspection Complete.
- 10. Aircraft forms Complete.
- 11. Aircraft secured Check.

N-17(N-18 Blank)



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EMERGENCY PROCEDURES

NOTE

The urgency of certain emergencies requires immediate and instinctive action by the pilot. The single most important consideration is aircraft control. All procedures are subordinate to this requirement.

ENGINE MALFUNCTION

ENGINE MALFUNCTION BEFORE LIFTOFF (ABORT)

- 1. Power levers IDLE.
- 2. Braking -- As Required.
- 3. Condition levers FUEL CUTOFF.
- 4. Fire pull handles PULL.
- 5. Master switch OFF.

ENGINE MALFUNCTION AFTER LIFTOFF (ABORT)

- 1. Power levers REDUCE.
- 2. Gear DOWN.
- 3. Condition levers FUEL CUTOFF.
- 4. Fire pull handles PULL.
- 5. Master switch OFF.





ENGINE MALFUNCTION AFTER LIFTOFF (FLIGHT CONTINUED)

- 1. Power MAXIMUM CONTROLLABLE.
- 2. <u>Gear UP.</u>
- 3. <u>Flaps UP.</u>
- 4. Landing lights OFF.
- 5. Brake deice OFF.
- 6. Engine cleanup PERFORM.
- 7. Generator load 100% MAXIMUM.

ENGINE MALFUNCTION AFTER LIFTOFF (FLIGHT CONTINUED WITHOUT AUTO-FEATHER)

- 1. Power Maximum controllable.
- 2. Dead engine Identify.
- 3. POWER lever (dead engine) IDLE.
- 4. PROP lever (dead engine) FEATHER.
- 5. <u>GEAR UP.</u>

- 6. <u>FLAPS UP.</u>
- 7. LANDING LIGHTS OFF.
- 8. BRAKE DEICE OFF.
- 9. Engine cleanup Perform.
- 10. Generator load 100% maximum.





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ENGINE MALFUNCTION DURING FLIGHT

- 1. Auopilot/yaw damp DISENGAGE.
- 2. <u>Power AS REQUIRED</u>.
- 3. <u>Dead engine IDENTIFY</u>
- 4. Power lever (affected engine) IDLE.
- 5. Propeller lever (affected engine) FEATHER.
- 6. <u>Gear AS REQUIRED</u>.
- 7. Flaps AS REQUIRED.
- 8. Power SET.
- 9. Engine cleanup PERFORM.
- 10. Generator Load 100% MAXIMUM.





ENGINE MALFUNCTION DURING FINAL APPROACH)

- 1. Power- AS REQUIRED.
- 2. Gear DN.

ENGINE MALFUNCTION (SECOND ENGINE)

- 1. Airspeed 140 KIAS.
- 2. Power lever IDLE.
- 3. Propeller lever DO NOT FEATHER.
- 4. Conduct engine restart procedure.

ENGINE SHUTDOWN IN FLIGHT

- 1. Power lever IDLE.
- 2. Propeller lever FEATHER.
- 3. Condition lever FUEL CUTOFF.
- 4. Engine cleanup PERFORM.

ENGINE CLEANUP

- 1. Condition lever FUEL CUTOFF.
- 2. Engine auto ignition switch OFF.
- 3. Autofeather switch OFF.
- 4. Generator switch OFF.
- 5. Prop sync switch OFF.
- 6. Radome heat OFF.





ENGINE RESTART DURING FLIGHT USING STARTER

- 1. Cabin temperature mode switch OFF.
- 2. Electrical load REDUCE TO MINIMUM.
- 3. Fire pull handle IN.
- 4. Power lever IDLE.
- 5. Propeller lever FEATHER.
- 6. Condition lever FUEL CUTOFF.
- 7. TGT (operative engine) 700° C OR LESS.
- 8. Ignition and engine start switch ON
- 9. Condition lever LOW IDLE.
- 10. TGT 1000° C 5 SECONDS MAXIMUM.
- 11. Oil pressure CHECK.

- 12. Ignition and engine start switch OFF.
- 13. Generator switch RESET, THEN ON.
- 14. Engine cleanup PERFORM IF ENGINE RESTART UNSUCCESSFUL.
- Cabin temperature mode switch AS REQUIRED.
- 16. Electrical equipment AS REQUIRED.
- 17. Auto ignition switch ARMED.
- 18. Propellers SYNCHRONIZED
- 19. Power AS REQUIRED.



ENGINE RESTART DURING FLIGHT (NOT USING STARTER)

- 1. Cabin temperature mode switch OFF.
- 2. Electrical load REDUCE TO MINIMUM.
- 3. Generator switch (affected engine) OFF.
- 4. Fire pull handle CHECK IN.
- 5. Power lever IDLE.
- 6. Propeller lever HIGH RPM.
- 7. Condition lever- FUEL CUTOFF.
- 8. Airspeed 140 KIAS minimum CHECK.
- 9. Altitude below 20,000 feet CHECK.
- 10. Engine auto ignition switch ARM.
- 11. Condition lever LOW IDLE.
- 12. TGT 1000°C 5 SECONDS MAXIMUM.
- 13. Oil pressure CHECK.
- 14. Generator switch RESET THEN ON.
- 15. Engine Cleanup PERFORM IF ENGINE RESTART UNSUCCESSFUL.
- 16. Cabin temperature mode switch AS REQUIRED.
- 17. Electrical equipment AS REQUIRED.
- 18. Auto ignition switch ARMED.
- 19. Propellers SYNCHRONIZED.
- 20. Power AS REQUIRED.

LOW OIL PRESSURE

- 1. Oil pressure below 105 PSI below 21,000 feet or 85 PSI 21,000 feet and above, torque 49% MAXIMUM.
- 2. Oil pressure below 60 PSI PERFORM ENGINE SHUTDOWN, OR LAND AS SOON AS PRACTICABLE USING MINIMUM POWER TO INSURE SAFE ARRIVAL.

DUCT OVERTEMP CAUTION LIGHT ILLUMINATED

- 1. Cabin air control IN.
- 2. Cabin temperature mode switch AUTO.
- 3. Cabin temperature rheostat FULL DECREASE.
- 4. Vent blower switch HI.
- 5. Cabin temperature mode switch MAN HEAT
- 6. Manual temperature switch DECREASE (HOLD).
- 7. Left bleed air valve switch ENVIRO OFF.
- 8. Light still illuminated (30 seconds) LEFT BLEED AIR VALVE SWITCH OPEN.
- 9. Right bleed air valve switch ENVIRO OFF.
- 10. Light still illuminated (30 seconds) RIGHT BLEED AIR VALVE OPEN.

ICE VANE FAILURE

- 1. Airspeed 160 KIAS OR BELOW.
- 2. Ice vane control circuit breaker PULL.
- 3. Ice vane OPERATE MANUALLY.
- 4. Airspeed RESUME NORMAL AIRSPEED.

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BLEED AIR FAILURE LIGHT ILLUMINATED

- 1. Brake deice switch OFF.
- 2. TGT and torque MONITOR (NOTE READINGS).
- 3. Bleed air valve switch PNEU & ENVIRO OFF.
- 4. Cabin pressurization CHECK.

EXCESSIVE DIRRERENTIAL PRESSURE

- 1. Cabin controller SELECT HIGHER SETTING.
- 2. If condition persists: LEFT BLEED AIR VALVE SWITCH ENVIRO OFF.
- 3. If condition still persists: RIGHT BLEED AIR VALVE SWITCH ENVIRO OFF.
- 4. If condition still persists: DESCEND IMMEDIATELY.
- 5. If unable to descend: CREW OXYGEN masks 100% AND ON.
- 6. If unable to descend: CABIN PRESS switch DUMP.
- 7. Bleed air valve switches OPEN.

LOSS OF PRESSURIZATION (ABOVE 10,000 FEET)

1. Crew oxygen masks - 100% AND ON.

CABIN DOOR CAUTION LIGHT ILLUMINATED.

- 1. Bleed air valve switches ENVIRO OFF.
- 2. Descend below 14,000 feet as soon as practicable.
- 3. Oxygen AS REQUIRED.

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SINGLE-ENGINE DESCENT/ARRIVAL

- 1. Cabin controller SET.
- 2. Seat belts and harnesses SECURE.
- 3. Ice and rain switches AS REQUIRED.
- 4. Altimeters SET.

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- 5. Recognition lights ON.
- 6. Arrival briefing COMPLETE.

SINGLE-ENGINE BEFORE LANDING

- 1. Propeller lever AS REQUIRED.
- 2. Flaps APPROACH.
- 3. Gear DN.
- 4. Landing lights AS REQUIRED.
- 5. Yaw damp OFF.
- 6. Brake deice OFF.

SINGLE-ENGINE LANDING CHECK

- 1. Autopilot/yaw damp DISENGAGED.
- 2. Gear lights CHECK.
- 3. Propeller lever (operative engine) HIGH RPM.

SINGLE-ENGINE GO-AROUND

- 1. Power MAXIMUM CONTROLLABLE.
- 2. Gear UP.
- 3. Flaps AS REQUIRED.
- 4. Landing light OFF.
- 5. Power AS REQUIRED.
- 6. Yaw damp AS REQUIRED.

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PROPELLER FAILURE (OVER 2080 RPM)

- 1. Power lever [affected engine IDLE.
- 2. Propeller lever FEATHER.
- 3 Condition lever AS REQUIRED.
- 4. Engine cleanup AS REQUIRED.



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FIRE

ENGINE / NACELLE FIRE DURING START OR GROUND OPERATIONS

- 1. Propeller levers FEATHER.
- 2. Condition levers FUEL CUTOFF.
- 3. Fire pull handle PULL.
- 4. Push to extinguish switch PUSH.
- 5. Master switch OFF.

ENGINE FIRE IN FLIGHT (FIRE PULL HANDLE LIGHT ILLUMINATED)

- 1. Power lever IDLE.
- 2. Fire pull handle light out ADVANCE POWER.

Fire pull handle light still illuminated - PERFORM ENGINE FIRE IN FLIGHT PROCEDURES IDENTIFIED).

ENGINE FIRE IN FLIGHT (IDENTIFIED)

- 1. Power lever IDLE.
- 2. Propeller lever FEATHER.
- 3. Condition lever FUEL CUTOFF.
- 4. Fire pull handle PULL.
- 5. Fire extinguisher- ACTIVE AS REQUIRED
- 6. Engine cleanup PERFORM.

FUSELAGE FIRE

- 1. Fight the fire.
- 2. Land as soon as possible.



WING FIRE

- 1. Perform engine shutdown on affected side.
- 2. Land as soon as possible.

ELECTRICAL FIRE

- 1. <u>Crew oxygen 100%.</u>
- 2. MASTER SWITCH OFF.
- 3. All nonessential electrical equipment OFF.
- 4. Battery switch ON.
- 5. Generator switches (individually) RESET, THEN ON.
- 6. Circuit breakers CHECK FOR INDICATION OF DEFECTIVE CIRCUIT.
- 7. Essential electrical equipment ON.
- 8. Land as soon as practicable.

SMOKE AND FUME ELIMINATION

- 1. Crew oxygen 100% AND ON.
- 2. Bleed air valve switches ENVIRO OFF.
- 3. Vent blower switch AUTO.
- 4. Aft vent blower switch OFF.
- 5. Cabin temperature mode switch OFF.
- 6. If smoke and fumes are not eliminated, CABIN PRESS switch DUMP.
- 7. Engine oil pressure MONITOR.

FUEL SYSTEM

FUEL PRESS WARNING LIGHT ILLUMINATED

- 1. Standby pump switch ON.
- 2. Fuel pressure light out CHECK.
- 3. Fuel pressure light still on RECORD UNBOOSTED TIME.

NO FUEL TRANSFER CAUTION LIGHT ILLUMINATED

- 1. AUX TRANSFER switch (affected side) -OVERRIDE.
- 2. Auxiliary fuel quantity MONITOR.
- 3. AUX TRANSFER switch AUTO.

NACELLE FUEL LEAK

- 1. Perform engine shutdown.
- 2. Fire pull handle PULL.
- 3. Land as soon as practicable.

FUEL CROSSFEED

- 1. AUX TRANSFER switches AUTO.
- 2. Standby pumps OFF.
- 3. Crossfeed switch AS REQUIRED.
- 4. Fuel crossfeed light illuminated CHECK.
- 5. Fuel pressure light extinguished CHECK.
- 6. Fuel quantity MONITOR.

NAC LOW LIGHT ILLUMINATED

- 1. Twenty minutes fuel remaining CONFIRM.
- 2. Land as soon as possible.

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ELECTRICAL SYSTEM

DC GEN LIGHT ILLUMINATED

- 1. Generator switch OFF, RESET, THEN ON.
- 2. Generator switch (no reset) OFF.
- 3. Mission control switch OVERRIDE.
- 4. Operating loadmeter 100% MAXIMUM.

BOTH DC GEN LIGHTS ILLUMINATED

- 1. All nonessential equipment OFF.
- 2. Land as soon as practicable.

EXCESSIVE LOADMETER INDICATION (OVER 100%)

- 1. Battery switch OFF (MONITOR LOADMETER).
- 2. Loadmeter over 100% NONESSENTIAL ELECTRICAL EQUIPMENT OFF.
- 3. Loadmeter under 100% BATT SWITCH ON.

INVERTER LIGHT ILLUMINATED

1. Affected AIRCRAFT INVERTER switch - OFF.

INST AC LIGHT ILLUMINATED

- 1. N_1 and TGT indications CHECK.
- 2. Other engine instruments MONITOR.

CICRUIT BREAKER TRIPPED

- 1. BUS FEEDER breaker tripped DO NOT RESET.
- 2. Nonessential circuit DO NOT RESET.
- 3. Essential circuit RESET ONCE.

BATTERY CHARGE LIGHT ILLUMINATED

- 1. Battery volt-amp meter CHECK.
- 2. Battery switch ON (FOR LANDING PRIOR TO GEAR AND FLAP EXTENSION).

EMERGENCY DESCENT

- 1. Power lever IDLE.
- 2. Propeller lever HIGH RPM.
- 3. Flap lever APPROACH.
- 4. Gear- DN.
- 5. Airspeed 184 KIAS MAXIMUM.

LANDING EMERGENCIES

LANDING GEAR UNSAFE INDICATION

- 1. LANDING GEAR RELAY circuit breaker -CHECK IN.
- 2. Gear lights CHECK.
- 3. Gear handle DN.
- 4. Manual gear extension AS REQUIRED.

LANDING GEAR EMERGENCY EXTENSION

- 1. Airspeed 130 KIAS.
- 2. LANDING GEAR RELAY circuit breaker OUT.
- 3. Gear handle DN.
- 4. Landing gear alternate engage handle LIFT AND TURN CLOCKWISE TO THE STOP.
- 5. Alternate landing gear extension handle -PUMP.
- 6. Gear lights illuminated CHECK.

GEAR-UP LANDING (ALL GEAR UP OR UNLOCKED)

- 1. Crew emergency briefing COMPLETE.
- 2. Loose equipment STOWED.
- 3. Bleed air valves ENVIRO OFF.
- 4. Cabin pressure switch DUMP.
- 5. Cabin emergency hatch REMOVE AND STOW.
- 6. Seat belts and harnesses SECURED.
- 7. Landing gear alternate engage handle DISENGAGED.
- 8. Alternate landing gear extension handle STOWED.
- 9. Gear relay circuit breaker IN.
- 10. Gear handle UP.
- 11. Nonessential electrical equipment OFF.
- 12. Flaps AS REQUIRED (DOWN FOR LANDING).
- 13. Condition levers FUEL CUTOFF.
- 14. Fire pull handles PULL.
- 15. Master switch OFF.

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LANDING WITH NOSE GEAR UNSAFE

- 1. Crew emergency briefing COMPLETE.
- 2. Loose equipment STOWED.
- 3. Bleed air valves ENVIRO OFF.
- 4. Cabin pressure switch DUMP.
- 5. Cabin emergency hatch REMOVE AND STOW.
- 6. Seat belts and harnesses SECURED.
- 7. Nonessential electrical equipment OFF.
- 8. Condition levers FUEL CUTOFF.
- 9. Fire pull handle PULL
- 10. Master switch OFF.

LANDING WITH ONE MAIN GEAR UNSAFE

- 1. Crew emergency briefing COMPLETE.
- 2. Loose equipment STOWED.
- 3. Bleed air valves ENVIRO OFF.
- 4. Cabin pressure switch DUMP.
- 5. Cabin emergency hatch REMOVE AND STOW.
- 6. Seat belts and harnesses SECURED.
- 7. Nonessential electrical equipment OFF.
- 8. Touchdown ON SAFE MAIN GEAR FIRST.
- 9. Condition levers FUEL CUTOFF.
- 10. Fire pull handle PULL
- 11. Master switch OFF.

CRACKED WINDSHIELD (INTERNAL CRACK)

- 1. Descend BELOW 25,000 FEET.
- 2. Cabin Pressure RESET PRESSURE DIFFERENTIAL TO 4 PSI OR LESS WITHIN 10 MINUTES.

CRACKED CABIN WINDOW

- 1. Oxygen AS REQUIRED.
- 2. Cabin pressure DEPRESSURIZE.
- 3. Descend AS REQUIRED.

DITCHING

- 1. Radio calls/transponder AS REQUIRED.
- 2. Crew emergency briefing AS REQUIRED.
- 3. Bleed air valves ENVIRO OFF.
- 4. Cabin pressurization switch DUMP.
- 5. Cabin emergency hatch REMOVE AND STOW.
- 6. Seat belts and harnesses SECURED.
- 7. Gear UP.
- 8. Flaps DOWN.
- 9. Nonessential electrical equipment OFF.
- 10. Approach NORMAL, POWER ON.
- 11. Emergency lights AS REQUIRED.

FLIGHT CONTROLS MALFUNCTION

UNSCHEDULED RUDDER BOOST ACTIVATION

1. Rudder boost - OFF.

IF CONDITION PERSISTS:

- 2. BLEED AIR VALVE (PNEU & ENVIRO) OFF.
- 3. Rudder trim ADJUST.

UNSCHEDULED ELECTRIC ELEVATOR TRIM

- 1. ELEV TRIM switch OFF.
- 2. ELEC TRIM circuit breaker OUT.

BAILOUT

- 1. Notify copilot to prepare to bail out.
- 2. Distress message TRANSMIT.
- 3. Voice security ZEROIZE.
- 4. Transponder 7700.
- 5. Flaps DOWN.
- 6. Airspeed 100 KIAS.
- 7. Trim AS REQUIRED.
- 8. Autopilot ENGAGE.
- 9. Cabin pressure switch DUMP.
- 10. Radio cord, oxygen hose, harnesses and seat belt -DISCONNECT.
- 11. Parachute ATTACH TO HARNESS.
- 12. Cabin door OPEN.
- 13. Abandon the aircraft.

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PERFORMANCE CHECKS

OXYGEN SYSTEM

- 1. Oxygen supply pressure gages Check.
- 2. Supply control lever (green) ON.
- 3. Diluter control lever 100% OXYGEN.
- 4. Emergency control lever (red) Set to TEST MASK position while holding mask directly away from face, then return to NORMAL.
- 5. Oxygen masks Don and adjust.
- 6. Emergency pressure control lever (red) Set to TEST MASK position and check mask for leaks, then return lever to NORMAL.
- 7. Flow indicator Check during inhalation blinker appears, during exhalation blinker disappears). Repeat a minimum of 3 times.

FUEL PUMPS/CROSSFEED OPERATION

- 1. Fire pull handles Pull.
- 2. Standby pump switches ON.
- 3. Battery switch ON.
- 4. #1 and #2 fuel press warning lights Illuminated.
- 5. Fire pull handles IN.
- 6. #1 and #2 fuel pressure warning lights -Extinguished.
- 7. Standby pump switches STANDBY PUMP.
- 8. #1 and #2 fuel pressure warning lights -Illuminated.

 Crossfeed - Check. Check system operation by activating switch momentarily left then right, noting that #1/#2 FUEL PRESS warning lights extinguish and that the FUEL CROSSFEED advisory light illuminates as switch is energized.

ANTI-ICE SYSTEMS - CHECK

- 1. Stall warning heat switch ON.
- 2. Pitot heat switches (2) ON. Check cover removed.
- 3. Fuel vent heat switches (2) ON.
- 4. Left wing heated fuel vent Check by feel for heat and condition.
- 5. Stall warning vane Check by feel for heat and condition.
- 6. Left pitot tube Check by feel for heat and free of obstructions.
- 7. Right pitot tube Check by feel for heat and free of obstructions.
- 8. Right wing heated fuel vent Check by feel for heat and condition.
- 9. Stall warning heat switch OFF.
- 10. Pitot heat switches (2) OFF.
- 11. Heated fuel vent switches (2) OFF.

ANNUNCIATOR PANELS

- MASTER CAUTION, MASTER WARNING, #1, FUEL PRESSURE, #2 FUEL PRESSURE, GEAR DN, L BL AIR FAIL, R BL AIR FAIL, ALT WARN, INST AC, #1 DC GEN, #1 INVERTER, #1 NO FUEL XFR, #2 NO FUEL XFR, #2 INVERTER, #2 DC GEN - Check on.
- 2. ANNUNCIATOR TEST switch Press and hold Check that all lights in aircraft and mission an

nunciator panels illuminate, FIRE PULL handle lights, marker beacon lights, MASTER CAUTION and MASTER WARNING lights are on. Release switch and check that all lights except those in step 1 are extinguished.

- 3. MASTER CAUTION and MASTER WARNING lights Press. Both lights extinguish.
- 4. Stall and gear warning system TEST. Check that warning horn sounds and that the LDG GEAR CONTR handle lights (2) illuminate.

FIRE PROTECTION SYSTEM

- Fire detector test switch Rotate counterclockwise to check three DETR positions. FIRE PULL handles should illuminate in each position. Reset MASTER WARNING in each position.
- Fire detector test switch Rotate counterclockwise to check two EXTGH positions. SQUIB OK light, associated EXTGH DISCH caution light and MASTER CAUTION LIGHT should illuminate in each position.

INERTIAL NAVIGATION SYSTEM ALIGNMENT

- 1. Exterior power 28V DC Connected.
- 2. Key lock switch ON.
- 3. Battery switch ON.
- 4. Aircraft inverters ON.
- 5. Mission inverters ON.
- 6. Mission control switch As required.
- 7. 3 phase A.C. bus RESET.

(Check inertial cooling for ON.)

8. Aircraft master avionics switch - EXTERNAL POWER.

- 9. Mode selector ALIGN.
- 10. Data selector DSTRK/STS.

(Align condition is shown on 5 digit RH display. Align will not progress beyond 8 until present position is loaded.)

11. Test button - PRESS AND HOLD.

All displays read 8, ROLL LIMIT, HOLD, INSERTS ADVANCE, WY PT, CHG, ALERT, BAT, WARN, and READY NAV LAMPS lit. When released, all extinguish except INSERT/ADVANCE. Insure all malfunction codes are cleared.

- 12. Data selector L/L POS (UTM for grid nav).
- 13. Load present position.
 - a. Select N or S degrees, minutes and tenths-INSERT/ADVANCE PRESS.
 - b. Select E or W degrees, minutes and tenths-INSERT/ADVANCE PRESS.

NOTE

Insure correct values for UTM grid spheroid coefficients are loaded when using UTM coordinates.

WAYPOINT SELECTION

- 1. Data selector L/L WY PT.
- 2. WYPT thumb wheel DESIRED WY PT (Do not use 0).
- 3. LAT/LONG waypoint (Deg., Min., Tenths) LOAD.
- 4. INSERT/ADVANCE PRESS AGAIN.

Latitude and longitude in arc-seconds relating to tenths entered is shown.

5. ARC/SEC LAT & LONG (Sec. and Tenths) LOAD.

Repeat 1 thru 5 for each WY PT.

 Flight plan cross check - Data selector to DIS/TIME (left display will indicate distance between WY PTS TO-FROM). Press WY PT change. Verify logical distance between waypoints.

TACAN STATION SELECTION

- 1. Data selector L/L WY PT.
- 2. Simultaneously press KEYS 7 and 9.
- 3. WY PT thumb wheel DESIRED TACAN.
- 4. TACAN station position LOAD LAT/LONG.
- 5. INSERT/ADVANCE PRESS.

Latitude and longitude in arc-seconds relating to tenths entered is shown.

- 6. ARC/SEC and Tenths LOAD.
- 7. INSERT/ADVANCE PRESS.
- 8. TACAN station altitude LOAD (Select KEY 4 OR 6). Enter altitude.
- 9. INSERT/ADVANCE PRESS
- 10. TACAN channel number LOAD (Select KEY 2 or 8). Enter channel.

Repeat steps 1 thru 8 for all TACAN stations.

HSI INTERFACE TEST

NOTE

Interface test must be performed after alignment progresses from state 8 but prior to switching to NAV.

- 1. AUTO/MAN switch MAN.
- 2. Couple INS to FD and engage AP.

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- 3. Data selector (Any position but DSRTK).
- 4. CDU TEST PRESS and HOLD.

MSU and CDU all lamps lit 8's. RMI all angles are 30 degrees. Cross track deviation bar is 1 dot right, NAV flags are retracted, WX radar NAV display indicates 1 dot right of course.

On HRI, a 15 degree RH steering command. Aircraft panel LINK UPDATE/TACAN UPDATE annunciated and INS light illuminated.

- Continue holding TEST switch from MAN to AUTO. RMI all angles are 0. Cross track deviation is one dot left. A 15 degree left steering command is issued.
- 6. Release TEST switch operations return to normal.

ELECTRIC ELEVATOR TRIM AND AUTOPILOT/FLIGHT DIRECTOR OPERATION

- Pilot and copilot PITCH TRIM switches Press to NOSE UP and NOSE DN positions, singularly and in pairs. Check that trim wheel moves in proper direction and operates only when trim switches are pressed in pairs. Any deviation requires that electric elevator trim be turned off and flight conducted using manual trim.
- 2. TRIM DISC switch Press and check that electric trim disconnects and that ELEV TRIM light extinguishes.
- 3. Flight director (FD) and radio magnetic indicators (RMI) warning flags masked Check.

NOTE

Since the pressure of airflow that normally opposes movement of control surfaces is absent during preflight check, it is possible to get a hard over control surface deflection if an autopilot command is allowed to remain active for any appreciable length of time. Move turn knob and pitch thumbwheel only enough to check operation, then return them to the center position.

- 4. Select HDG mode Check.
- 5. Horizontal situation indicator (HSI) heading marker under lubber line Set.
- 6. Engage autopilot and check controls stiff and AIL HI TORQUE, HDG, and AP ENG are illuminated Check.
- 7. Move HSI heading marker 100 left and right and verify that FD and control wheels respond in the appropriate direction Check.
- 8. Press AP/YD disengage switch and verify that autopilot disengages and that flight controls are free -Check.
- 9. Engage autopilot Check.
- 10. Command 50 trim UP with AP pitch wheel and verify that manual trim wheel moves nose UP and AP trim light indicates UP trim Check.
- 11. Press pitch trim switch nose down and verify that autopilot disengages and AUTOPILOT TRIM FAIL and MASTER WARNING lights illuminate Check.

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NOTE

The AP TRIM FAIL annunciator will extinguish by pressing the AP/YD disconnect button on the control wheel to the first detent.

- 12. Repeat steps 9 thru 11 above using opposite commands.
- 13. Engage autopilot Check.
- 14. Move HSI heading marker to command a bank on Flight Director Check.
- 15. Press GO-AROUND switch and verify that GA annunciator light illuminates, autopilot disengages, and that flight director commands a wings level, 7° nose-up attitude Check.
- 16. Press TEST switch (pilot's HRI) and verify that attitude display indicates an additional 10° pitch up and 20° right bank Check.
- 17. Engage autopilot command DN with AP pitch wheel and engage and hold AUTO PILOT TRIM TEST switch when elevator trim wheel starts to rotate.
- 18. Verify that autopilot disengages and AP TRIM FAIL and MASTER WARNING lights illuminate within 10 seconds.

AVIONICS CHECKS

NAV 1

- 1. Frequency select knob (NAV panel) Select VOR frequency.
- 2. NAV TEST switch (NAV PANEL) Press and hold.
- 3. RMI Observe that single needle indicates approximately 005
- 4. VOR/LOC flag Check that flag is out of view.

- 5. TO/FROM pointer Check that pointer indicates TO.
- 6. HSI course deviation bar Check for centered bar.
- 7. Marker beacon lights Check that all three lamps are illuminated and flickering at approximately a 30 Hz rate.
- 8. VOR frequency knob (NAV panel) Select a LOC frequency.
- 9. HSI course deviation bar Check that bar indicates a deflection of approximately one dot right of center.
- 10. HSI glideslope pointer Check that pointer indicates a deflection of approximately one dot below center.
- 11. Marker beacon lights Check that all three lamps are illuminated and flickering at approximately a 30 Hz rate.

NAV 2

All NAV 2 self-test procedures are the same as those used for NAV 1, with the exception of the marker beacon test. There is no marker beacon receiver in the NAV 2 system.

TACAN

- 1. TEST pushbutton Press and hold.
- 2. Range indicator Check for an indication of 0.0 ± 0.1 nautical miles.
- 3. Pilot's COURSE SELECTOR switch Select TACAN.
- 4. Pilot's RMI selector switch Select TACAN.
- 5. RMI double needle Check for an indication of $180^{\circ} \pm 20$.

- HSI course selector Turn to 180° and adjust slowly until the course deviation bar is centered The bar should center between a selected course of 178° to 182°.
- HSI course selector Turn the selector +10° from the setting achieved in step 6, and check that course deviation bar is located over the far left 10° dot.
- 8. HSI course selector Turn the selector +10° from the setting
- HSI course selector Turn the selector+10° from the setting achieved in step 6, and check that course deviation bar is located over the far right 10° dot.
- 10. TO-FROM indicator Check that TO is indicated.
- 11. TEST pushbutton Release.

AUTOFEATHER

- 1. Condition levers LOW IDLE.
- 2. Autofeather switch Hold to TEST.
- 3. Power levers Advance until AUTOFEATHER lights are illuminated (approximately 22% torque).
- 4. #1 power lever Retard.
- a. At approximately 18% torque #2 AUTOFEATHER light out.
- b. At approximately 12% torque Both AU, TOFEATHER lights out (propeller starts tog feather).
- 5. #1 power lever Approximately 22% torque.
- 6. Repeat steps 2 thru 4 for #2 engine.

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OVERSPEED GOVERNORS

- 1. Power levers Set approximately 1950 RPM (both engines).
- 2. #1 propeller governor test switch Hold.
- 3. #1 propeller RPM 1830 to 1910 Check.
- 4. Repeat steps 2 and 3 for #2 engine.
- 5. Power levers Set 1800 RPM.

PRIMARY GOVERNORS

- 1. Power 1800 RPM Set/check.
- 2. Propeller levers aft to detent Set.
- 3. Propeller RPM 1600 1640 Check.
- 4. Propeller levers to HIGH RPM Set.

CE VANES

- 1. Ice vane switches to EXTEND. Verify torque drop, TGT increase, and illumination of ICE VANE EXT light -Check.
- 2. Ice vane switches to RETRACT. Verify return to original torque and TGT and ICE VANE light extinguished -Check.

ANTI-ICE AND DEICE SYSTEMS

- 1. Left pitot switch ON Check for loadmeter rise, then OFF.
- 2. Right pitot switch ON Check for loadmeter rise, then OFF.
- 3. Stall warning switch ON Check for loadmeter rise, then OFF.
- 4. Fuel vent switch ON Check for loadmeter rise, then OFF.

- 5. Windshield anti-ice switches NORMAL and HI -Check PILOT and COPILOT (individually) foxy loadmeter rise, then OFF.
- 6. Propeller AUTO (Check 14 to 18 amps).
- 7. Propeller switches INNER and OUTER (momentarily), check for loadmeter rise.
- 8. Surface deice switch AUTO Check for a drops in pneumatic pressure and wing deice boots inflation and after 6 seconds for a second drop in pneumatic pressure. Check manual position for proper indications.
- 9. Antenna deice single cycle auto Check for drop in pneumatic pressure and boots inflated Check manual position for proper indications.
- 10. Radome anti-ice ON, check for proper indications.
- 11. Engine inlet lip heat switches ON, check for proper indications.
- 12. Anti-ice and deice systems switches As required.

PNEUMATIC PRESSURE

- 1. Condition levers HIGH IDLE.
- 2. Power levers IDLE.
- 3. Left bleed air valve switch PNEU & ENVIRO OFF.
- 4. Pneumatic pressure 12-20 PSI Check.
- 5. Left bleed air light Check illuminated.
- 6. Right bleed air valve switch PNEU & ENVIRO OFF.
- 7. Left and right bleed air off and left and right bleed air fail lights Check illuminated.

- 8. Left bleed air valve switch OPEN.
- 9. Left bleed air off, and left and right bleed air fail lights off, and pneumatic pressure Check (12 to 20 PSI).
- 10. Right bleed air valve switch OPEN.
- 11. Right bleed air off light Extinguished.

PRESSURIZATION SYSTEM

- 1. Cabin door caution light extinguished Check.
- 2. Storm windows closed Check.
- 3. Bleed air valve switches OPEN Check.
- 4. Cabin altitude 500 feet lower than field pressure altitude Set.
- 5. Cabin pressurization switch TEST (hold).
- 6. Cabin climb gage descending indication -Check, then release TEST switch.
- 7. Aircraft altitude set to planned cruise altitude plus 500 feet Check (if this setting does not result in a CABIN ALT indication of at least 500 feet over takeoff field pressure altitude, adjust as required).
- 8. Rate control set between 9 and 12 o'clock -Check.

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TM 55-1510-219-CL

By Order of the Secretary of the Army:

CARL E. VUONO General, United States Army Chief of Staff

Official:

PATRICIA P. HICKERSON Colonel, United States Army The Adjutant General

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The Metric System and Equivalents

Linear Measure

1 centimeter = 10 millimeters = .39 inch 1 decimeter = 10 centimeters = 3.94 inches 1 meter = 10 decimeters = 39.37 inches 1 dekameter = 10 meters = 32.8 feet 1 hectometer = 10 dekameters = 328.08 feet 1 kilometer = 10 hectometers = 3,280.8 feet

Weights

1 centigram = 10 milligrams = .15 grain 1 decigram = 10 centigrams = 1.54 grains 1 gram = 10 decigram = .035 ounce 1 decagram = 10 grams = .35 ounce 1 hectogram = 10 decagrams = 3.52 ounces 1 kilogram = 10 hectograms = 2.2 pounds 1 quintal = 100 kilograms = 220.46 pounds 1 metric ton = 10 quintals = 1.1 short tons

Liquid Measure

- 1 centiliter = 10 milliters = .34 fl. ounce
- 1 deciliter = 10 centiliters = 3.38 fl. ounces
- 1 liter = 10 deciliters = 33.81 fl. ounces
- 1 dekaliter = 10 liters = 2.64 gallons
- 1 hectoliter = 10 dekaliters = 26.42 gallons
- 1 kiloliter = 10 hectoliters = 264.18 gallons

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